Mars Exploration Program

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4/7/18

Mars Exploration Program Science Goals

Life

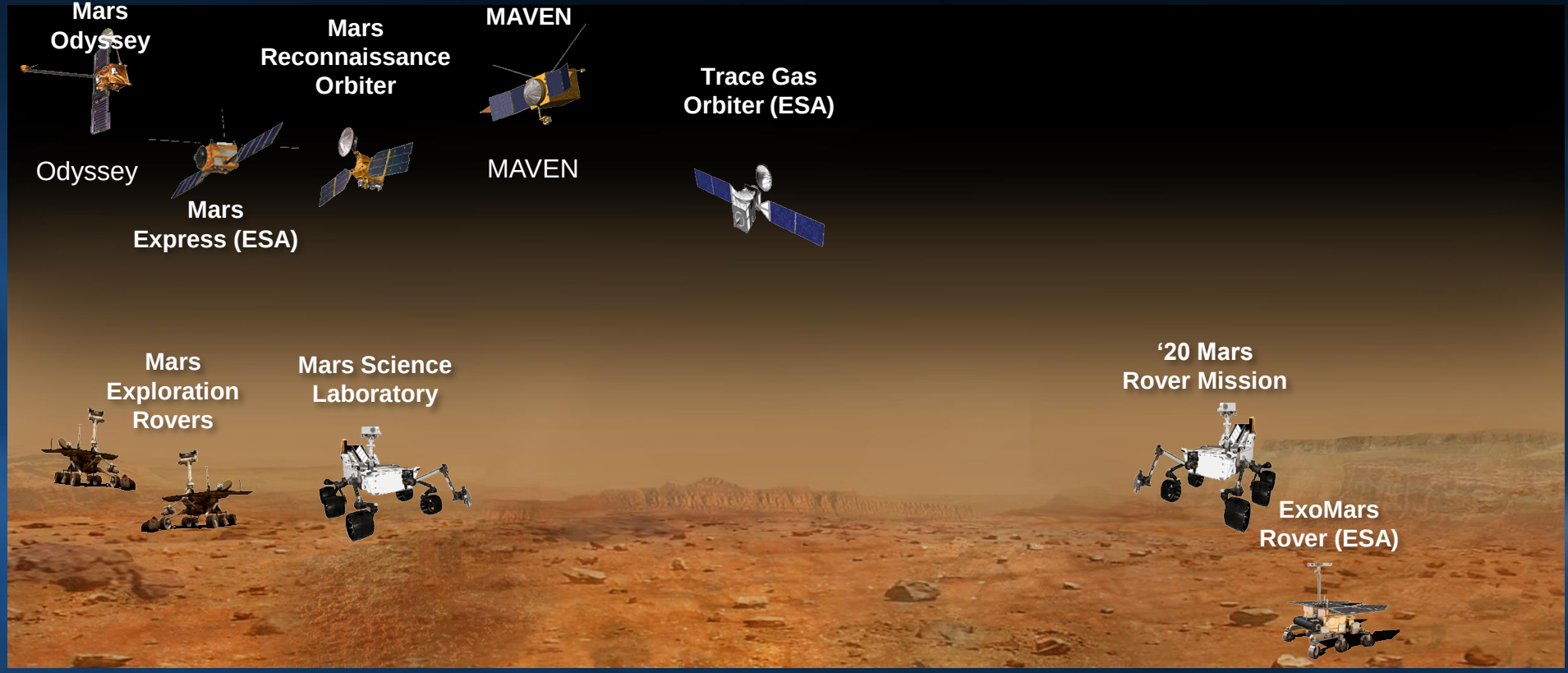
Climate

Geology & Geophysics

Prepare for Human Exploration

Mars Exploration Program Missions

2001 - 2015 | 2016 | 2018 | 2020 | Future Mars Missions

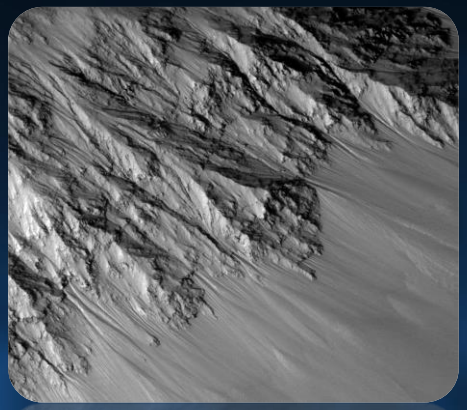


Mars Exploration Program Highlights

Opportunity: Journey to Perseverance Valley

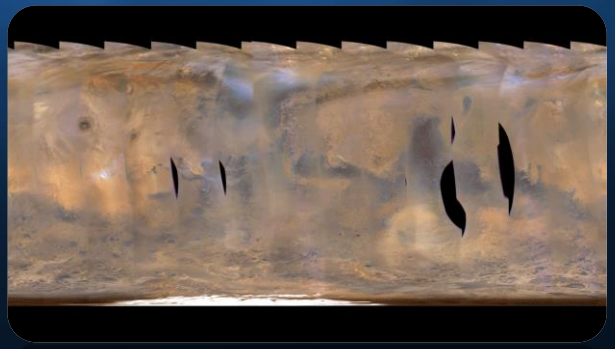
MRO: >50,000 orbits Completed Global 6m Resolution Imagery

MRO: Continuing Observations of Recurring Slope Lineae



Curiosity: >5 years since landing

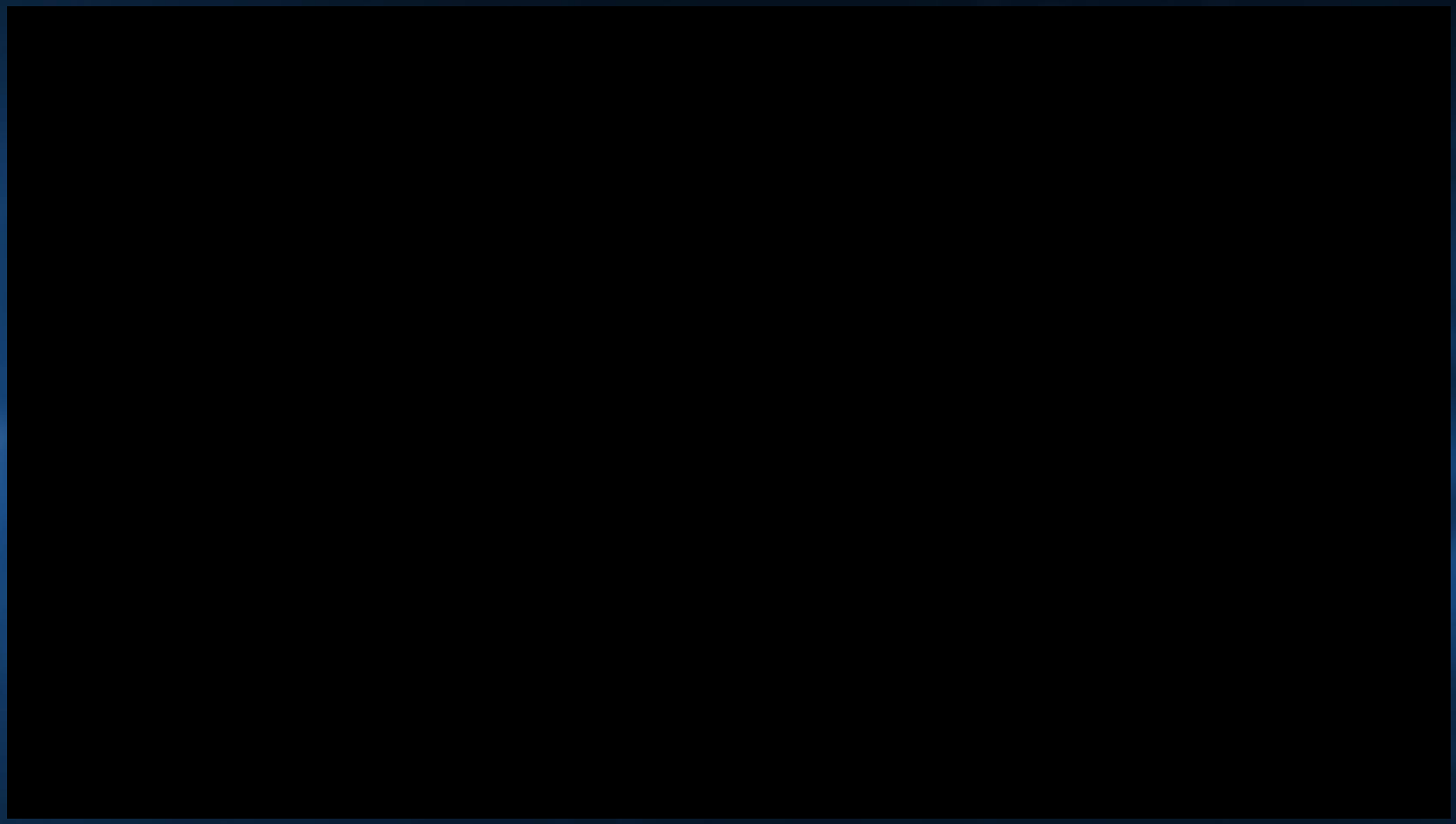
MAVEN Tracks Back-to-back Regional Storms



Mars 2020 Landing Site Finalists

Program accomplishments

Curiosity Landing video



Mars 2020 Status

Mars 2020 - Seeking the Signs of Life on Mars

Mars 2020 Mission Objectives

A. Geologic History

Explore the geology of the landing site

B. In Situ Astrobiology

Seek the signs of possible *ancient* Martian life

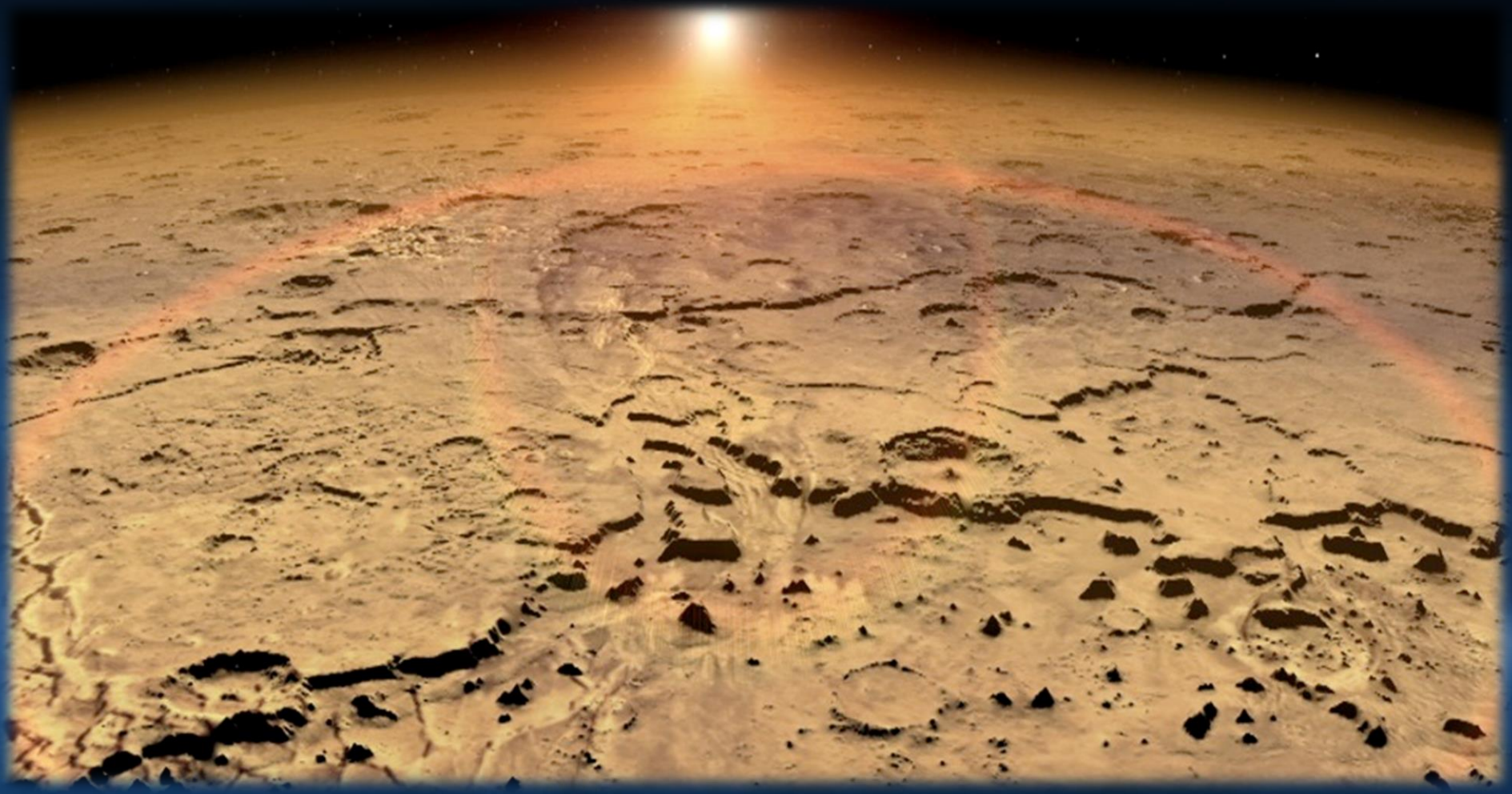
C. Cache samples

Prepare samples for possible future return to Earth

D. Prepare for the future

Demonstrate/test technologies required for human exploration of Mars

Why look for *ancient* life on Mars?

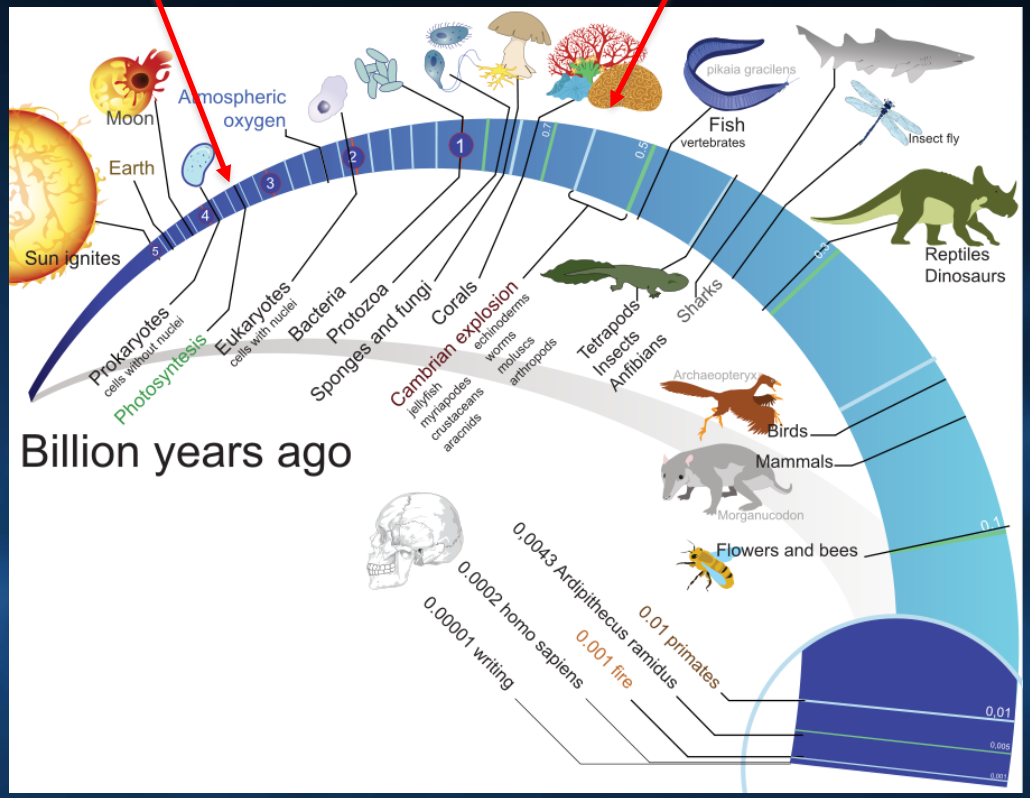


Mars is now cold, dry, and there is almost nowhere to hide from radiation

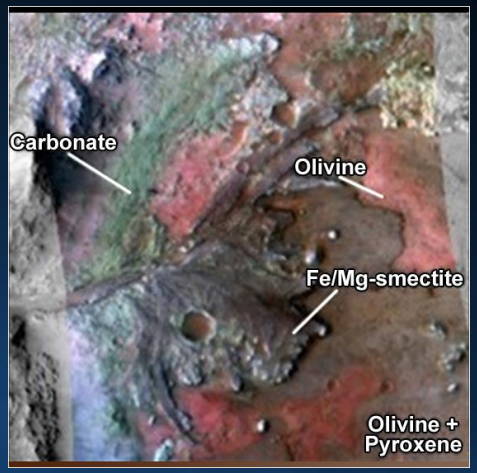
Mars life, if any, is expected to be microbial…

Mars climate change here

Familiar "fossils"



Mars 2020 Candidate Landing Sites



JEZERO

- Minerologic diversity including clays and carbonates
- Shallow water carbonates?

NE SYRTIS

- Extremely ancient igneous, hydrothermal, and sedimentary environments
- High mineralogic diversity with phyllosilicates, sulfates, carbonates, olivine

COLUMBIA HILLS

- Carbonate, sulfate, and silica-rich outcrops of possible hydrothermal origin. Hesperian volcanics.
- Previously explored by MER

Mars 2020: Mission Overview

LAUNCH

- Atlas V 541 vehicle
- Launch Readiness Date: July 2020
- Launch window: July/August 2020

CRUISE/APPROACH

- ~7 month cruise
- Arrive Feb 2021

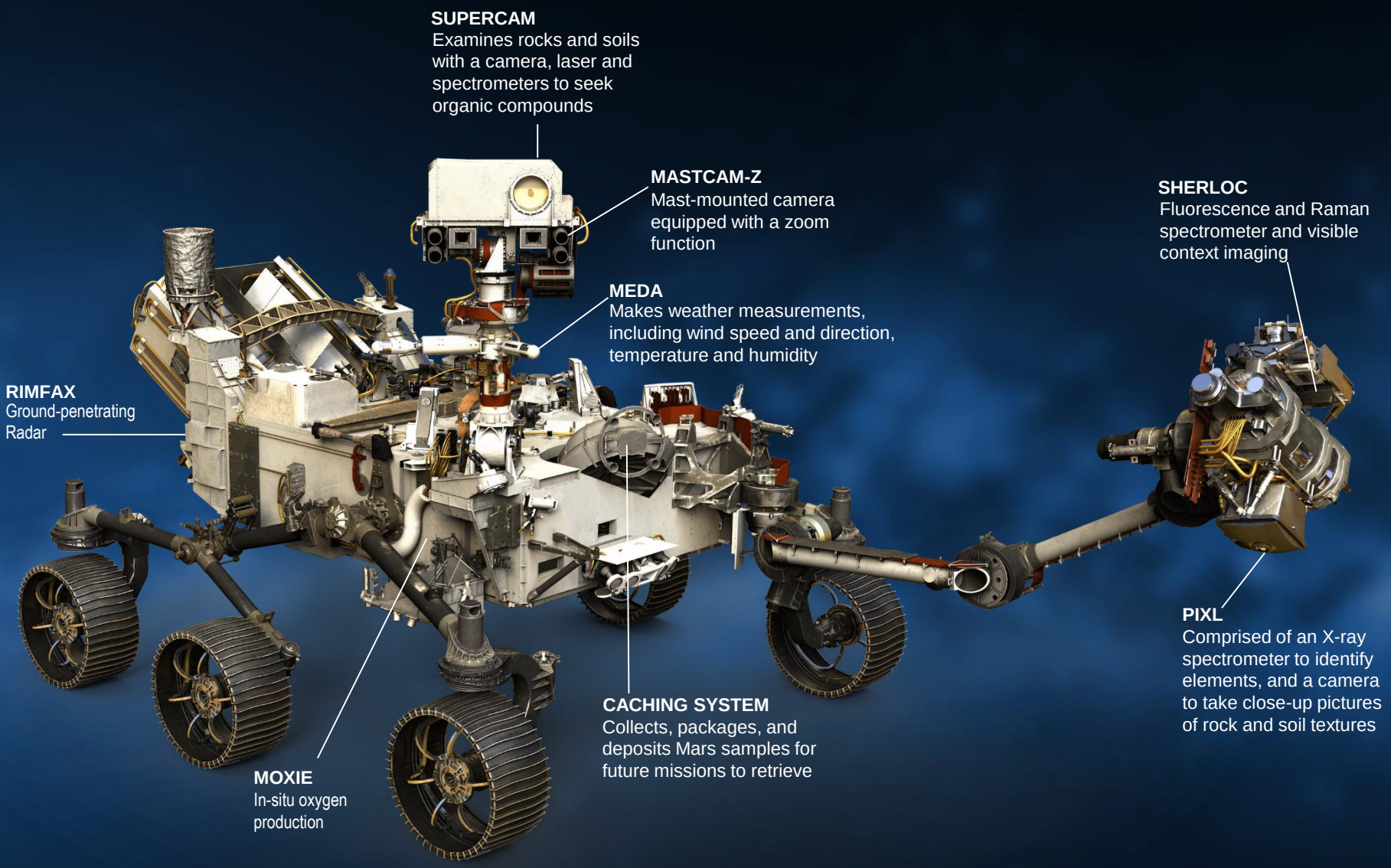
ENTRY, DESCENT & LANDING

- MSL EDL system (+ Range Trigger and Terrain Relative Navigation): guided entry and powered descent/Sky Crane
- 16 x 14 km landing ellipse (range trigger baselined)
- Access to landing sites ±30° latitude, ≤ -0.5 km elevation
- Curiosity-class Rover

SURFACE MISSION

- 20 km traverse distance capability
- Enhanced surface productivity
- Qualified to 1.5 Martian year lifetime
- Seeking signs of past life
- Returnable cache of samples
- Prepare for human exploration of Mars

Mars 2020 Rover Instruments



Descent Stage Walkdown 1/9/18

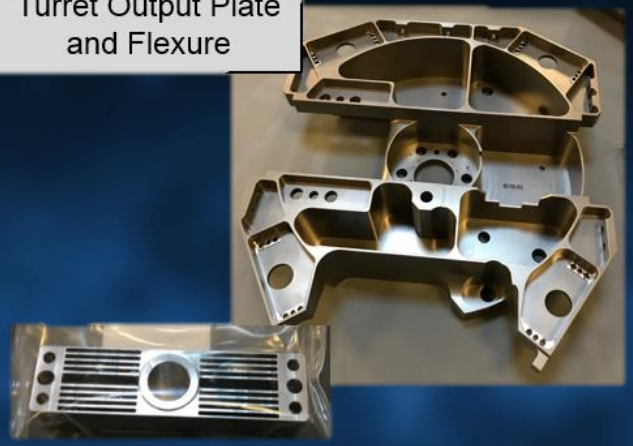
Cruise Stage Radiator Bakeout

Robotic Arm Highlights

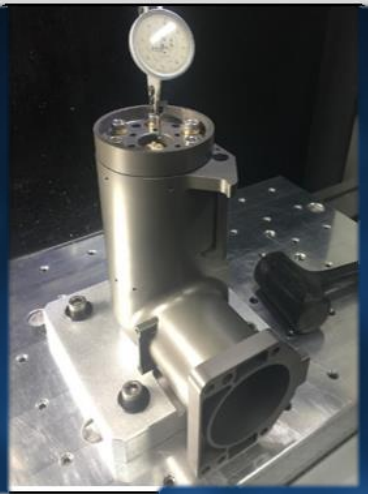
Piece Part Fabrication Continues

Upper Arm Fittings 80% complete

Turret Output Plate and Flexure



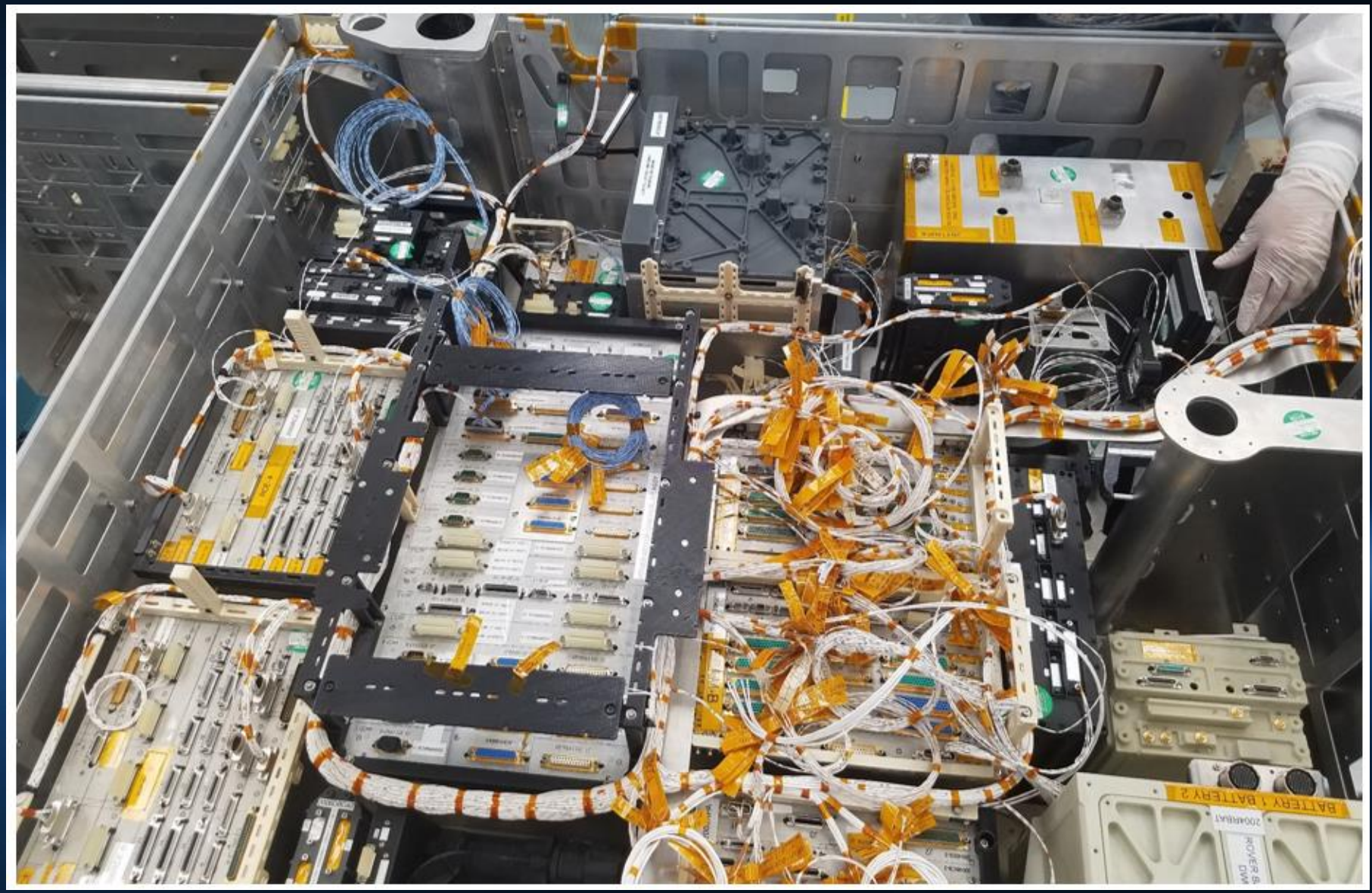
Wrist/turret fitting being match machined to facilitate wrist and turret actuator assembly



Turret Output shaft caps arrived

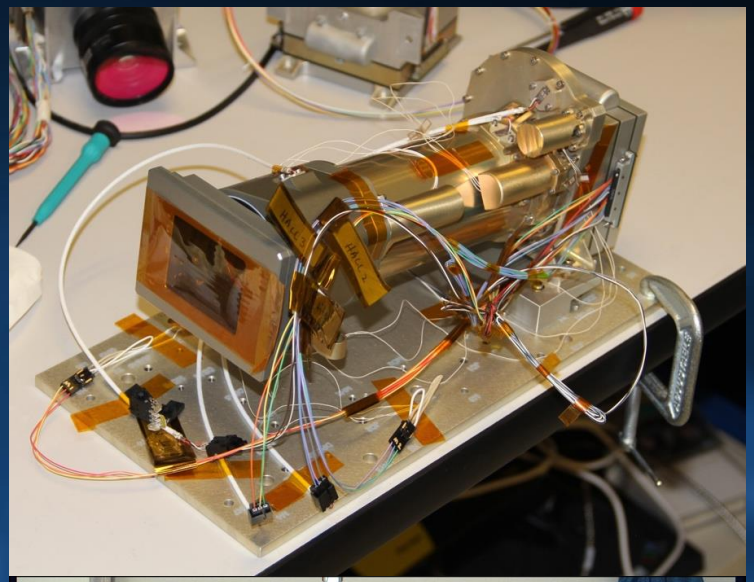
EM Shoulder Bracket 50% complete

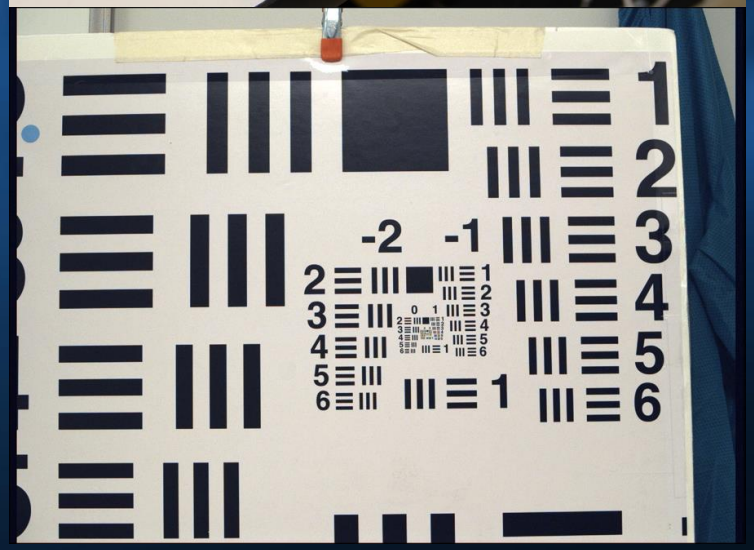
Rover Harness Production



Payload
Mastcam-Z Hardware

Assembled Mastcam-Z EQM at MSSS
Image date: 23 January, 2018





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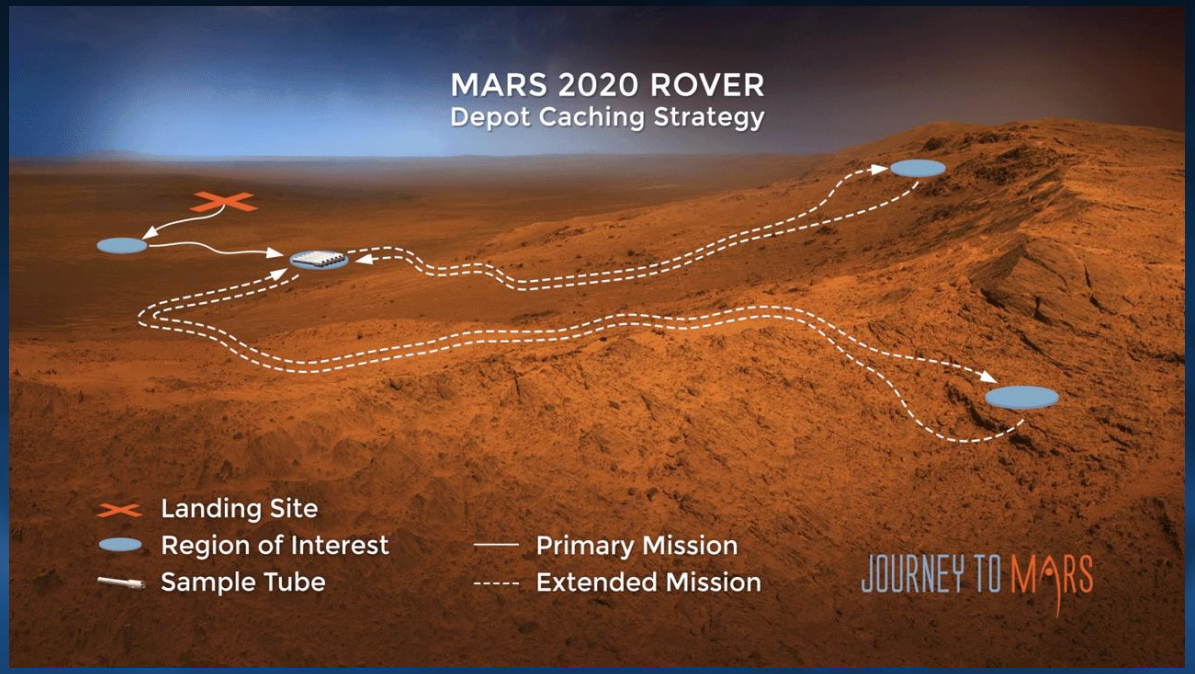
MARS 2020 Project



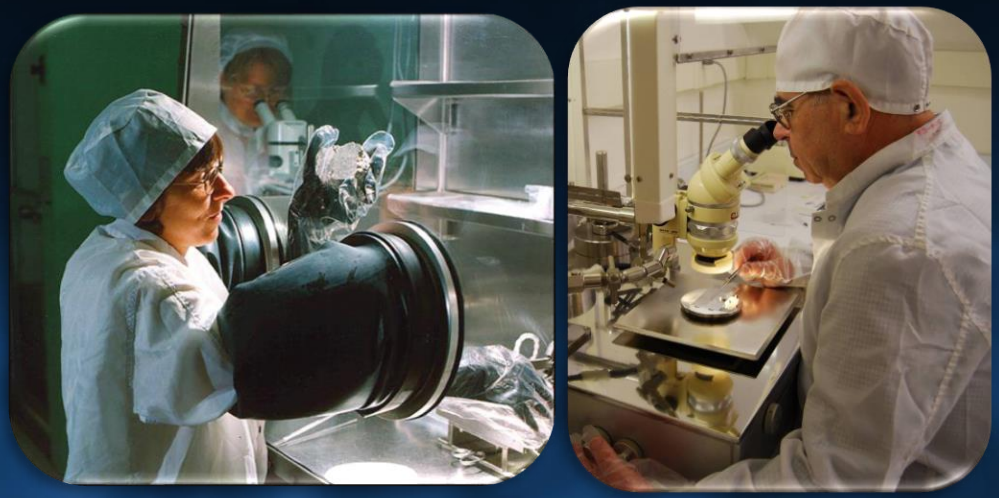
Why Mars Sample Return?

There are three primary reasons why MSR is of such high value to science.

1. Complex sample preparation, decisions

2. Better Instrumentation

3. Instrument Diversity



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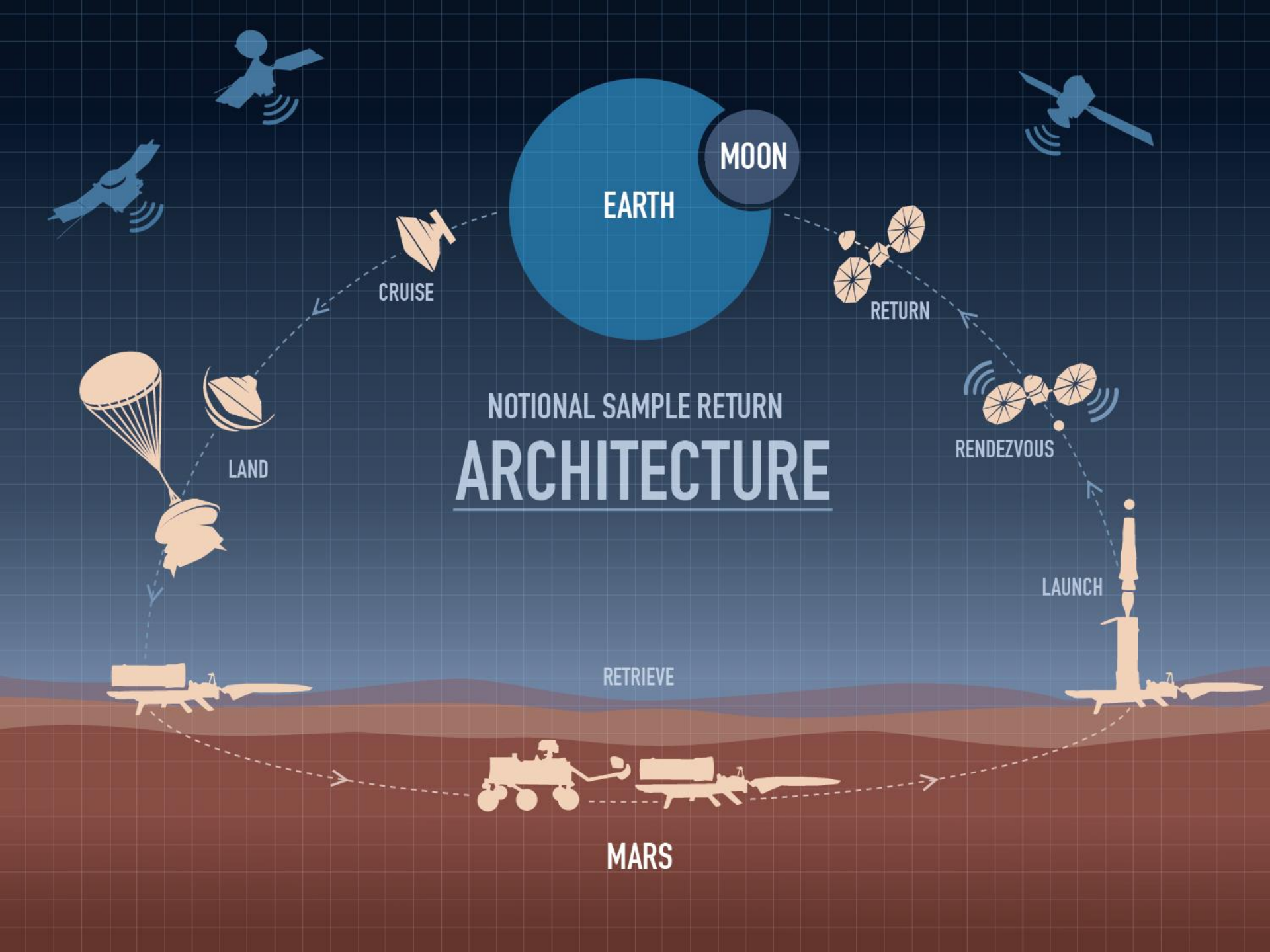
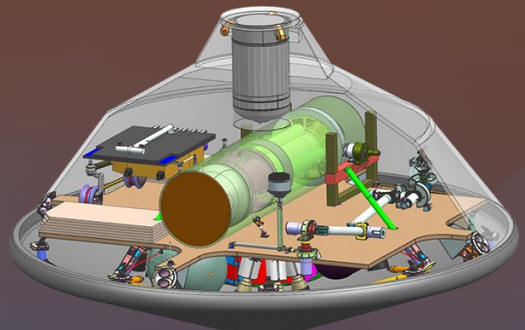
In-situ missions limited to a few sample analysis instruments.

Mars Sample Return Planning

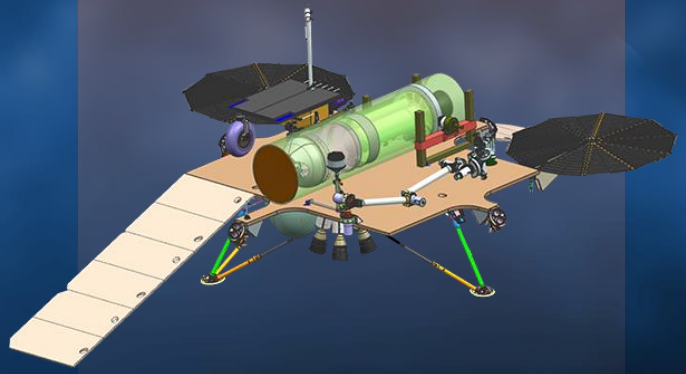

MOON
EARTH
CRUISE
RETURN
NOTIONAL SAMPLE RETURN
ARCHITECTURE
LAND
RENDEZVOUS
LAUNCH
RETRIEVE
MARS

TWO LANDER CONCEPTS

2017 Highly Integrated Concept



Propulsive Platform Lander (PPL) *Concept Packaged in MSL 4.5m Aeroshell*

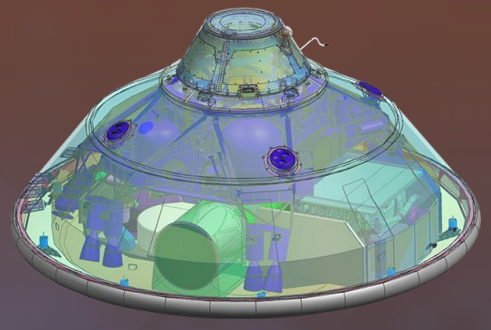


Propulsive Platform Lander
Concept Deployed

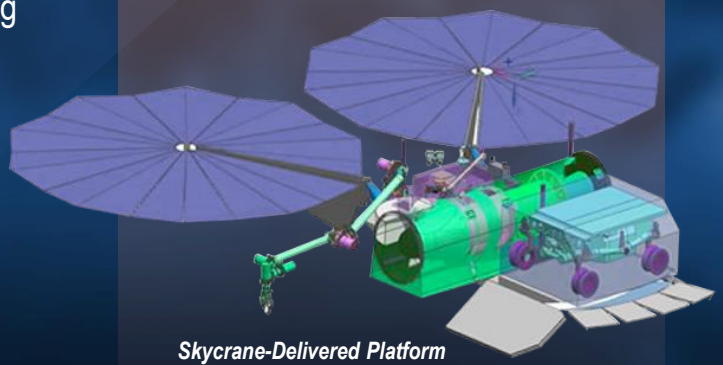
Common Attributes

- Identical cruise and entry architecture
- ~ 10 km landing ellipse
- ~ 900-1000 kg landed useful mass
- Accommodates ~ 600 kg MAV and Fetch Rover

Evolved 2011 Decadal Concept



Skycrane-Delivered Platform Concept Packaged in MSL 4.5m Aeroshell



Skycrane-Delivered Platform
Concept Deployed

Two concepts that leverage Mars program legacy system capabilities

NOTIONAL SAMPLE RETURN ORBITER

Design for Orbital Rendezvous & Fast Sample Return

- Rendezvous & Capture
- Containment and Earth Planetary Protection
- Communication Relay Support for Surface Ops and Critical Events
- Return to Earth, either via
  - Direct return to Earth
  - Deliver to cis-lunar space for human-assisted returns

Implementation Options

- NASA provided
- Partner provided